FLUID DYNAMICS MAT 401 UNIT 1



Dr. Rajshekhar Roy Baruah

Velocity Potential

Let \vec{q} be the fluid velocity at any instant *t* then the equations of the streamline at that instant is

$$\frac{dx}{u} = \frac{dy}{v} = \frac{dz}{w}$$

These curves cut the surfaces

udx + vdy + wdz = 0

orthogonally.

Let us consider a scalar function $\phi(x, y, z, t)$ at that instant, uniform throughout the entire field such that

$udx + vdy + wdz = -d\phi$

$$= \operatorname{vd} x + \operatorname{vd} y + \operatorname{wd} z = -\frac{\partial \phi}{\partial x} dx - \frac{\partial \phi}{\partial y} dy - \frac{\partial \phi}{\partial z} dz$$

efore,
$$u = -\frac{\partial \phi}{\partial y} \qquad u = -\frac{\partial \phi}{\partial y} \qquad u = -\frac{\partial \phi}{\partial y}$$

Ther

 $| = > \vec{q} = -\nabla \phi = -\text{grad}\phi$

 $\partial x'$

where ϕ is termed as the velocity potential for the field \vec{q} . The negative sign is taken as the matter of convention. This shows that ϕ decreases with an increase in the value of x, y or z i.e. the flow is always in the direction of decreasing ϕ .

dy

 ∂z

It ensures that the flow takes place from the higher to lower potentials. The velocity potential is a scalar function of space and time.

The necessary and sufficient condition for $\vec{q} = -\nabla \phi$ to hold is

 $\nabla imes \vec{q} = 0$

$$=>\hat{\imath}\left(\frac{\partial w}{\partial y}-\frac{\partial v}{\partial z}\right)+\hat{\jmath}\left(\frac{\partial u}{\partial z}-\frac{\partial w}{\partial x}\right)+\hat{k}\left(\frac{\partial v}{\partial x}-\frac{\partial u}{\partial y}\right)=0$$

If the above relation exists then the flow is said to be irrotational. In other words when the motion is irrotational the velocity vector is the gradient of a scalar function $\phi(x, y, z, t)$.

Remark 1. The surfaces $\phi(x, y, z, t) = \text{constant}$ are called the equipotentials. The streamlines

 $\frac{dx}{u} = \frac{dy}{v} = \frac{dz}{w}$ are cut at right angles by the surfaces given by the equation $\frac{dx}{u} = \frac{dy}{v} = \frac{dz}{w}$

and the condition for the existence of such orthogonal surfaces is the condition that udx + vdy + wdz = 0 may posses a solution of the form $\phi(x, y, z, t) = \text{constant}$ at the considered instant *t*, the analytical condition being

$$u\left(\frac{\partial w}{\partial y} - \frac{\partial v}{\partial z}\right) + v\left(\frac{\partial u}{\partial z} - \frac{\partial w}{\partial x}\right) + w\left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}\right) = 0$$

$$u = -\frac{\partial \phi}{\partial x}, \quad v = -\frac{\partial \phi}{\partial y}, \quad w = -\frac{\partial \phi}{\partial z}$$

When the velocity potential exists, then the above equation holds. Therefore,

$$\frac{\partial w}{\partial y} - \frac{\partial v}{\partial z} = \frac{\partial}{\partial y} \left(-\frac{\partial \phi}{\partial z} \right) - \frac{\partial}{\partial z} \left(-\frac{\partial \phi}{\partial y} \right) = -\frac{\partial^2 \phi}{\partial y \partial z} + \frac{\partial^2 \phi}{\partial z \partial y} = 0$$

=> $\frac{\partial w}{\partial y} = \frac{\partial v}{\partial z}$
fimilarly,
 $\frac{\partial u}{\partial z} = \frac{\partial w}{\partial x}, \quad \frac{\partial v}{\partial x} = \frac{\partial u}{\partial y}$

S

Remark 2. When $\nabla \times \vec{q} = 0$ holds, the flow is known as the potential kind. It is also known as irrotational. For such flow the field of \vec{q} is conservative.

Remark 3. The equation of continuity of an incompressible fluid is $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0$

Let the fluid move irrotationally. Then the velocity potential ϕ exists such that Such that $u = -\frac{\partial \phi}{\partial x}, \quad v = -\frac{\partial \phi}{\partial y}, \quad w = -\frac{\partial \phi}{\partial z}$ Therefore we get, $\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} + \frac{\partial^2 \phi}{\partial z^2} = 0$

Thus ϕ is a harmonic function satisfying the Laplace equation $\nabla^2 \phi = 0$, where

$$\nabla^2 \equiv \frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} + \frac{\partial^2}{\partial z^2}$$

Vorticity Vector

Let $\vec{q} = u\hat{\imath} + v\hat{\jmath} + w\hat{k}$ be the fluid velocity such that $\vec{q} \neq 0$. Then the vector $\vec{\Omega} = curl \vec{q}$ is called the *vorticity vector*. Let, Ω_x , Ω_y , Ω_z be the components of $\vec{\Omega}$ in cartesian coordinates.

Then

$$\Omega_x \hat{\imath} + \Omega_y \hat{\jmath} + \Omega_z \hat{k} = \hat{\imath} \left(\frac{\partial w}{\partial y} - \frac{\partial v}{\partial z} \right) + \hat{\jmath} \left(\frac{\partial u}{\partial z} - \frac{\partial w}{\partial x} \right) + \hat{k} \left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right)$$
so that

so that,

$$\Omega_x = \frac{\partial w}{\partial y} - \frac{\partial v}{\partial z}, \quad \Omega_y = \frac{\partial u}{\partial z} - \frac{\partial w}{\partial x}, \quad \Omega_z = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$$

Remark 1: In two dimensional cartesian coordinates, the vorticity is given by $\Omega_z = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$

Remark 2: In two dimensional polar coordinates, the vorticity is given by

$$\Omega_z = \frac{v_\theta}{r} + \frac{\partial v_\theta}{\partial r} - \frac{1}{r} \frac{\partial v_r}{\partial \theta}$$

Vortex Lines

A vortex line is a curve drawn in the fluid such that the tangent to it at every point is in the direction of the vorticity vector $\vec{\Omega}$.

Let, $\vec{\Omega} = \Omega_x \hat{\imath} + \Omega_y \hat{\jmath} + \Omega_z \hat{k}$ and $\vec{r} = x\hat{\imath} + y\hat{\jmath} + z\hat{k}$ be the position vector of a point P on a vortex line. Then $\vec{\Omega}$ is parallel to $d\vec{r}$ at P on the vortex line. Hence the equation of vortex lines is given by $\vec{\Omega} \times d\vec{r} = 0$

 $=> (\Omega_x \hat{\iota} + \Omega_y \hat{\jmath} + \Omega_z \hat{k}) \times (dx \hat{\iota} + dy \hat{\jmath} + dz \hat{k}) = 0$ $=> (\Omega_y dz - \Omega_z dy) \hat{\iota} + (\Omega_z dx - \Omega_x dz) \hat{\jmath} + (\Omega_x dy - \Omega_y dx) \hat{k} = 0$

$$=>\Omega_y dz - \Omega_z dy = 0, \ \Omega_z dx - \Omega_x dz = 0, \ \Omega_x dy - \Omega_y dx = 0$$

$$=> \frac{dx}{\Omega_x} = \frac{dy}{\Omega_y} = \frac{dz}{\Omega_z}$$

which gives the equations of vortex lines.



Vortex Jube and Vortex Filament

If we draw the vortex lines from each point of a closed curve in the fluid we obtain a tube called the *vortex tube*.

A vortex tube of infinitesimal cross-section is known as *vortex filament*.





Rotational and Irrotational Motion

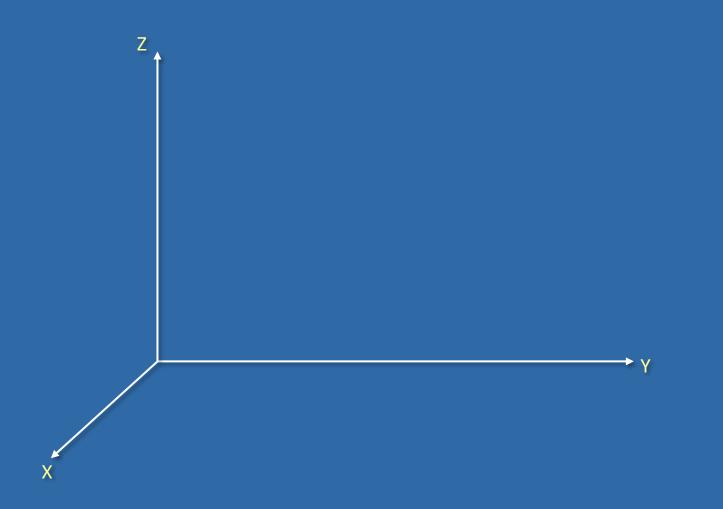
The motion of a fluid is said to be *irrotational* when the vorticity vector $\vec{\Omega}$ of every fluid particle is zero. When the vorticity vector is different from zero, the motion is said to be *rotational*.

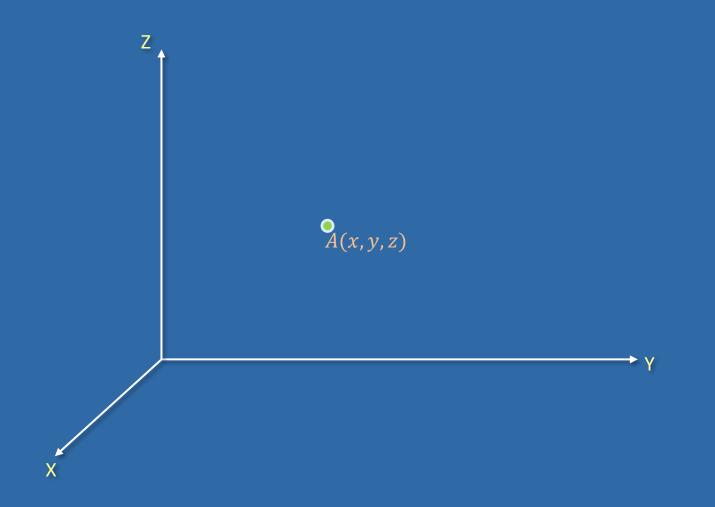
When the motion is *irrotational* i.e. when $\operatorname{curl} \vec{q} = 0$, then \vec{q} must be of the form $(-\operatorname{grad} \phi)$ for some scalar point function ϕ (say) because $\operatorname{curl} \operatorname{grad} \phi = 0$. Thus velocity potential exists whenever the fluid motion is irrotational. Again when velocity potential exists, the motion is irrotational because $\vec{q} = -\operatorname{grad} \phi = \operatorname{curl} \vec{q} = \operatorname{curl} \operatorname{grad} \phi = 0$

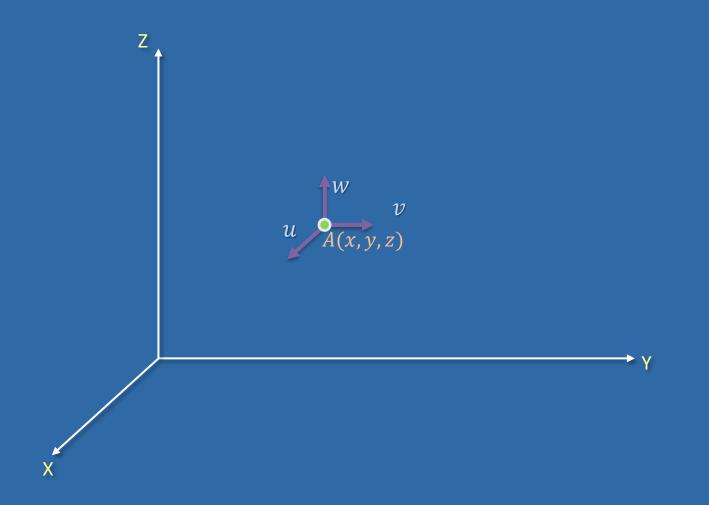
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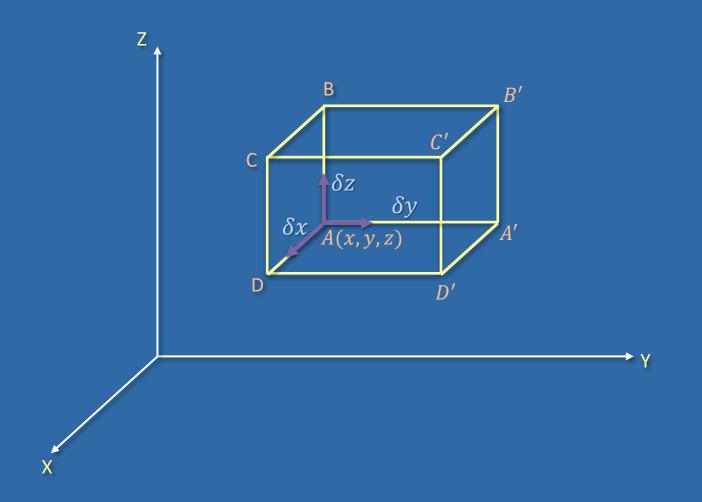


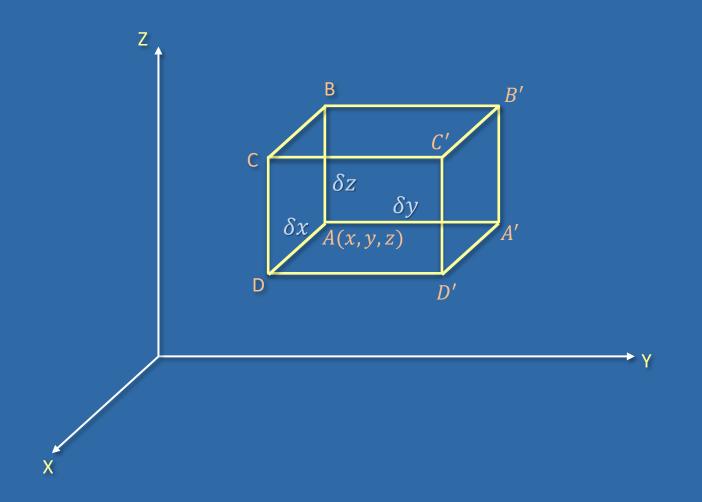
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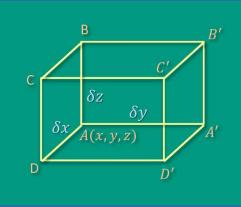






Let,

- Fluid particle at A(x, y, z).
- $\triangleright \rho$ be the density and p be the pressure.
- *u*, *v*, *w* be the velocity components at *A* parallel to the rectangular coordinate axes.
- (X, Y, Z) be the components of external force per unit mass at time t.



Let us construct a small parallelepiped with edges δx , δy , δz of lengths parallel to their respective coordinate axes, having A at one of the angular points as shown in the figure.

$Force = Pressure \times Area$

Force due to pressure on the face $ABCD = p\delta y\delta z = f(x, y, z)$ (1)

Force due to pressure on the face $A'B'C'D' = f(x + \delta x, y, z)$

$$= f(x, y, z) + \delta x \frac{\partial}{\partial x} f(x, y, z) + \cdots$$
$$= f(x, y, z) + \delta x \frac{\partial}{\partial x} f(x, y, z) \qquad (2)$$

Resultant force due to pressure along x - axis is

(2)-(1) =
$$-\delta x \frac{\partial}{\partial x} f(x, y, z) = -\frac{\partial p}{\partial x} \delta x \delta y \delta z$$

 \succ Du/Dt is the total acceleration of the element in x-direction.

- > The mass of the element is $\rho \delta x \delta y \delta z$.
- > The external force on the element in x-direction is $X\rho\delta x\delta y\delta z$.

By *Newton's second law of motion*, the equation of motion in x-direction is

Mass × (acceleration in x-direction) = Sum of the components of external forces in x-direction

$$i. e. \rho \delta x \delta y \delta z \frac{Du}{Dt} = X \rho \delta x \delta y \delta z - \frac{\partial p}{\partial x} \delta x \delta y \delta z$$

or,
$$\frac{Du}{Dt} = X - \frac{1}{\rho} \frac{\partial p}{\partial x}$$
(4)

Similarly, the equations of motion in *y* and *z*-directions are, respectively

$$\frac{Dv}{Dt} = Y - \frac{1}{\rho} \frac{\partial p}{\partial y}$$
(5)
$$\frac{Dw}{Dt} = Z - \frac{1}{\rho} \frac{\partial p}{\partial z}$$
(6)

$$\frac{D}{Dt} = \frac{\partial}{\partial t} + u\frac{\partial}{\partial x} + v\frac{\partial}{\partial y} + w\frac{\partial}{\partial z}$$

Rewriting (4), (5) and (6) the so-called Euler's dynamical equations of motion in cartesian coordinates are

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z} = X - \frac{1}{\rho} \frac{\partial p}{\partial x}$$
(7)
$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z} = Y - \frac{1}{\rho} \frac{\partial p}{\partial y}$$
(8)
$$\frac{\partial w}{\partial t} + u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z} = Z - \frac{1}{\rho} \frac{\partial p}{\partial z}$$
(9)

Multiplying equations (4) by \hat{i} , (5) by \hat{j} and (6) by \hat{k} we get,

$$\frac{Du}{Dt}\hat{\imath} = X\hat{\imath} - \frac{1}{\rho}\frac{\partial p}{\partial x}\hat{\imath}$$

$$\frac{Dv}{Dt}\hat{j} = Y\hat{j} - \frac{1}{\rho}\frac{\partial p}{\partial y}\hat{j}$$

$$\frac{Dw}{Dt}\hat{k} = Z - \frac{1}{\rho}\frac{\partial p}{\partial z}\hat{k}$$

Adding all the three equations we get,

$$\frac{D}{Dt}(u\hat{\imath} + v\hat{\jmath} + w\hat{k}) = (X\hat{\imath} + Y\hat{\jmath} + Z\hat{k}) - \frac{1}{\rho} \left[\frac{\partial}{\partial x}\hat{\imath} + \frac{\partial}{\partial y}\hat{\jmath} + \frac{\partial}{\partial z}\hat{k}\right]p$$

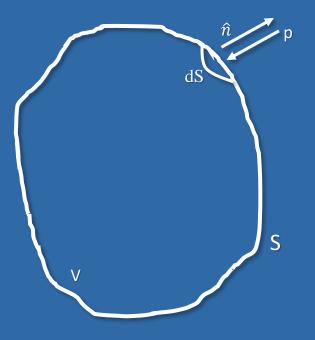
$$=>\frac{D\vec{q}}{Dt}=\vec{F}-\frac{1}{\rho}\nabla p$$

which is *Euler's equation of motion*.

FLUID DYNAMICS MAT 401 UNIT 1



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Euler's Equation

By Newton's second law of motion,

The total force acting on the mass of the fluid = rate of change of momentum

Total force = surface force + body force

Momentum = mass × velocity = density × volume ×velocity

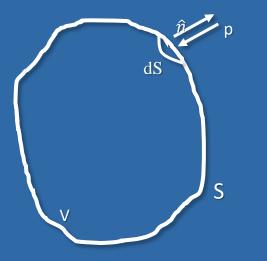
Mass = density × volume = ρdV Momentum = velocity × mass = $\vec{q}\rho dV$ Total momentum, $\vec{M} = \int_{V} \vec{q}\rho dV$ Rate of change of momentum = $\frac{d}{dt}(\vec{M})$

 $\frac{d\vec{M}}{dt} = \frac{d}{dt} \int_{V} \vec{q} \rho dV = \int_{V} \frac{d\vec{q}}{dt} \rho dV + \int_{V} \vec{q} \frac{d}{dt} (\rho dV)$ $\frac{d\vec{M}}{dt} = \int_{V} \frac{d\vec{q}}{dt} \rho dV \qquad [$

[Since mass is constant]

Euler's Equation

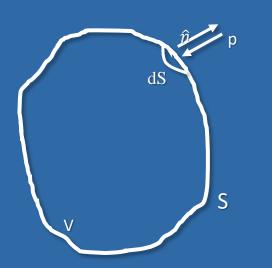
Force = pressure × area



Surface force on dS = $pdS(-\hat{n})$

Euler's Equation

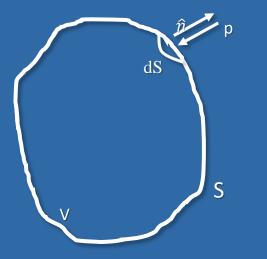
Force = pressure \times area



Total surface force on
$$S = \int_{S} p dS(-\hat{n})$$

= $-\int_{S} p \hat{n} dS$
= $-\int_{V} \nabla p dV$

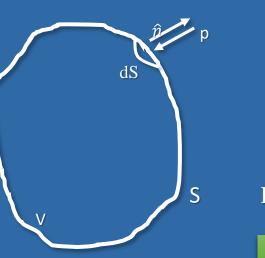
Force = pressure × area



Body force on $dS = \vec{F}\rho dV$

Euler's Equation

Force = pressure \times area



Total body force on
$$S = \int_{V} \vec{F} \rho dV$$

$$\therefore \text{ Total force} = \int_{V} \vec{F} \rho dV - \int_{V} \nabla p dV$$

From Newton's second law of motion,

Total force = Rate of change of momentum

$$\int\limits_{V} \frac{d\vec{q}}{dt} \rho dV = \int\limits_{V} \vec{F} \rho dV - \int\limits_{V} \nabla p dV$$

$$=>\int_{V}\left[\frac{d\vec{q}}{dt}\rho-\vec{F}\rho+\nabla p\right]dV=0$$

$$=>\frac{d\vec{q}}{dt}\rho-\vec{F}\rho+\nabla p=0$$

$$=>\frac{d\vec{q}}{dt}=\vec{F}-\frac{1}{\rho}\nabla p$$

which is *Euler*'s equation of motion.

Euler's Equation

